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A Numerical Solver  
for the  
National Cycle Program

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## **NCP**

### **National Cycle Program**

- The NCP is a NASA/Industry Cooperative Effort to build a common system modeling capability for aerospace propulsion systems.
- *General Electric Aircraft Engines (Cincinnati)*
- *Pratt & Whitney (East Hartford, CT)*
- *Boeing (Seattle, WA)*
- *Allied Signal (Phoenix, AZ)*
- *Arnold Engineering Development Center. (TN)*

- The NCP Solver is designed to solve 0-D to 1-D system simulation problems.
- Physical systems are decomposed into discrete elements (inlet, compressor, burner, etc.).
- Elements are connected in a network.
- NCP Solver manages execution of elements and data flow between them.
- The level of detail in each element can vary (from data tables through full 3D models) but at the system level, everything is 1-D.

## NCP Solver Flexibility

- The NCP is more flexible than most commercially available simulation codes, which were designed primarily for controls design.
- Any model input parameter can be controlled by the Solver in order to achieve a desired goal.
- The NCP solver can handle any problem that can be phrased as

**vary  $x_1$  until  $a_1 = b_1$**

- There can be more than one parameter  $x$  controlling an equal number of balances between  $a$  and  $b$ .

**vary  $\{x_1, x_2, \dots, x_n\}$  until  $\{(a_1=b_1) \& (a_2=b_2) \& \dots \& (a_n=b_n)\}$**

# NCP

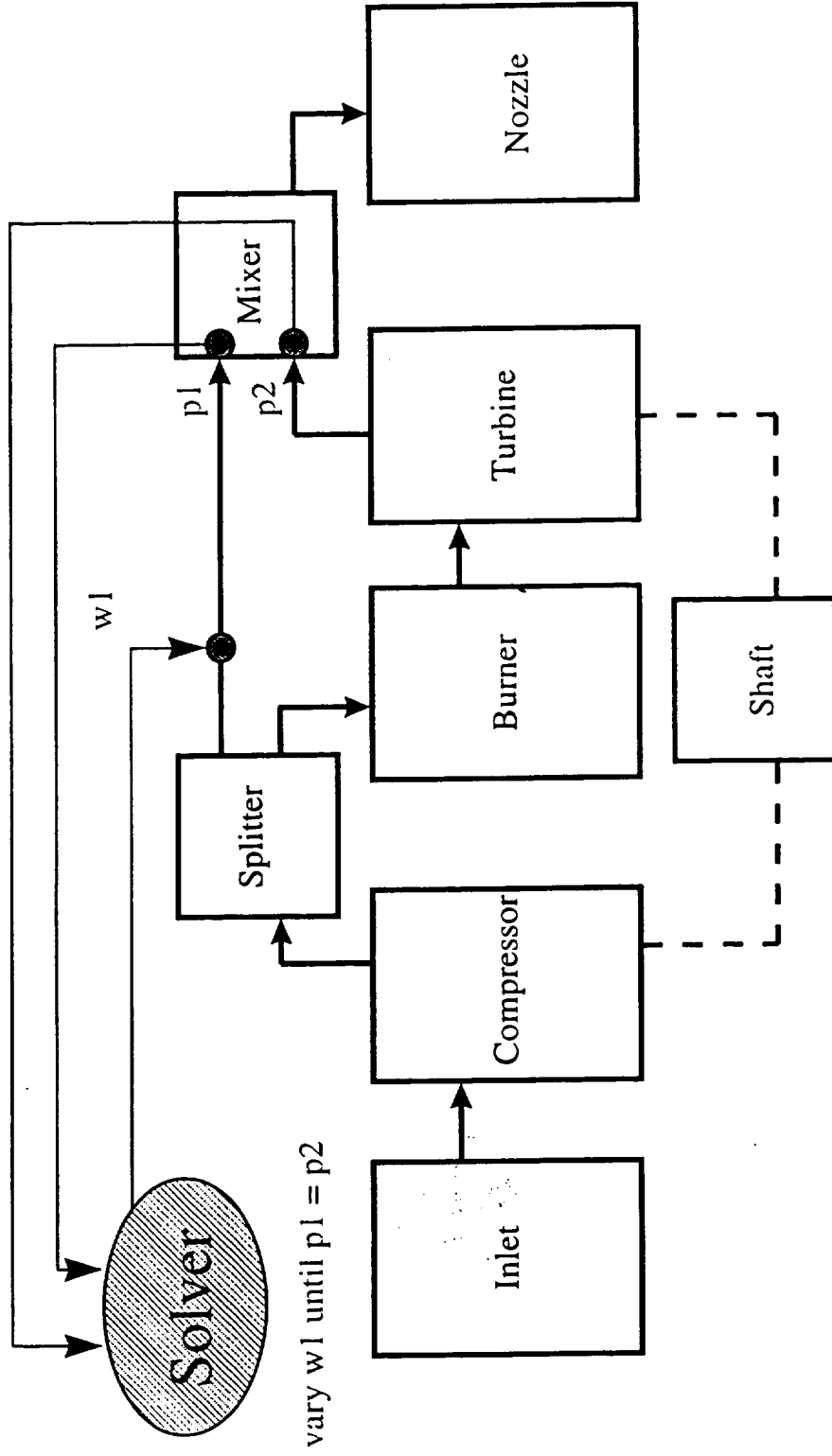
Analysis

Design

Data Reduction

Transient

Constraints



*Analysis*

# NCP

Analysis

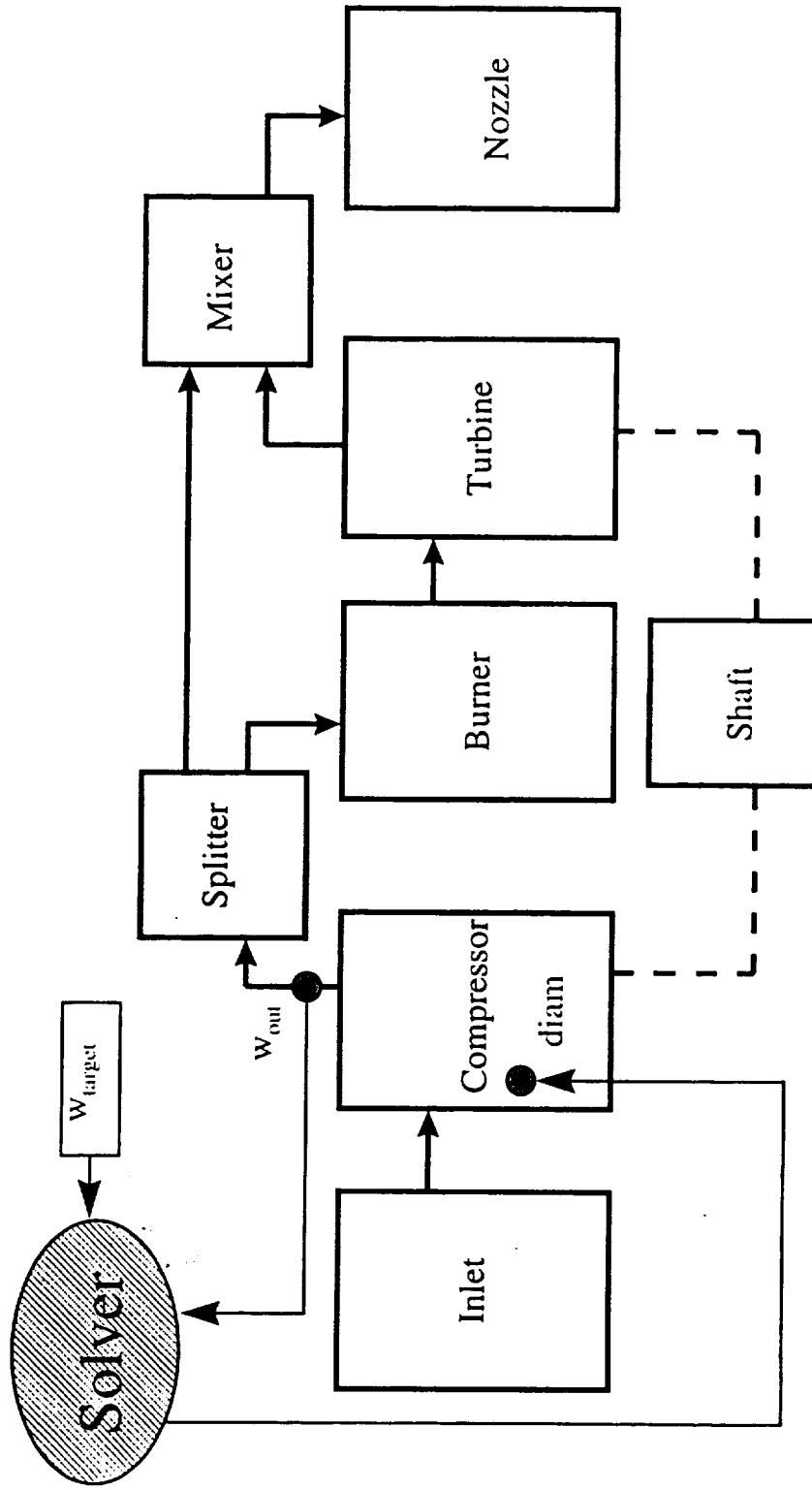
Design

Data Reduction

Transient

Constraints

vary diam until  $w_{out} = w_{target}$



*Design*



# NCP

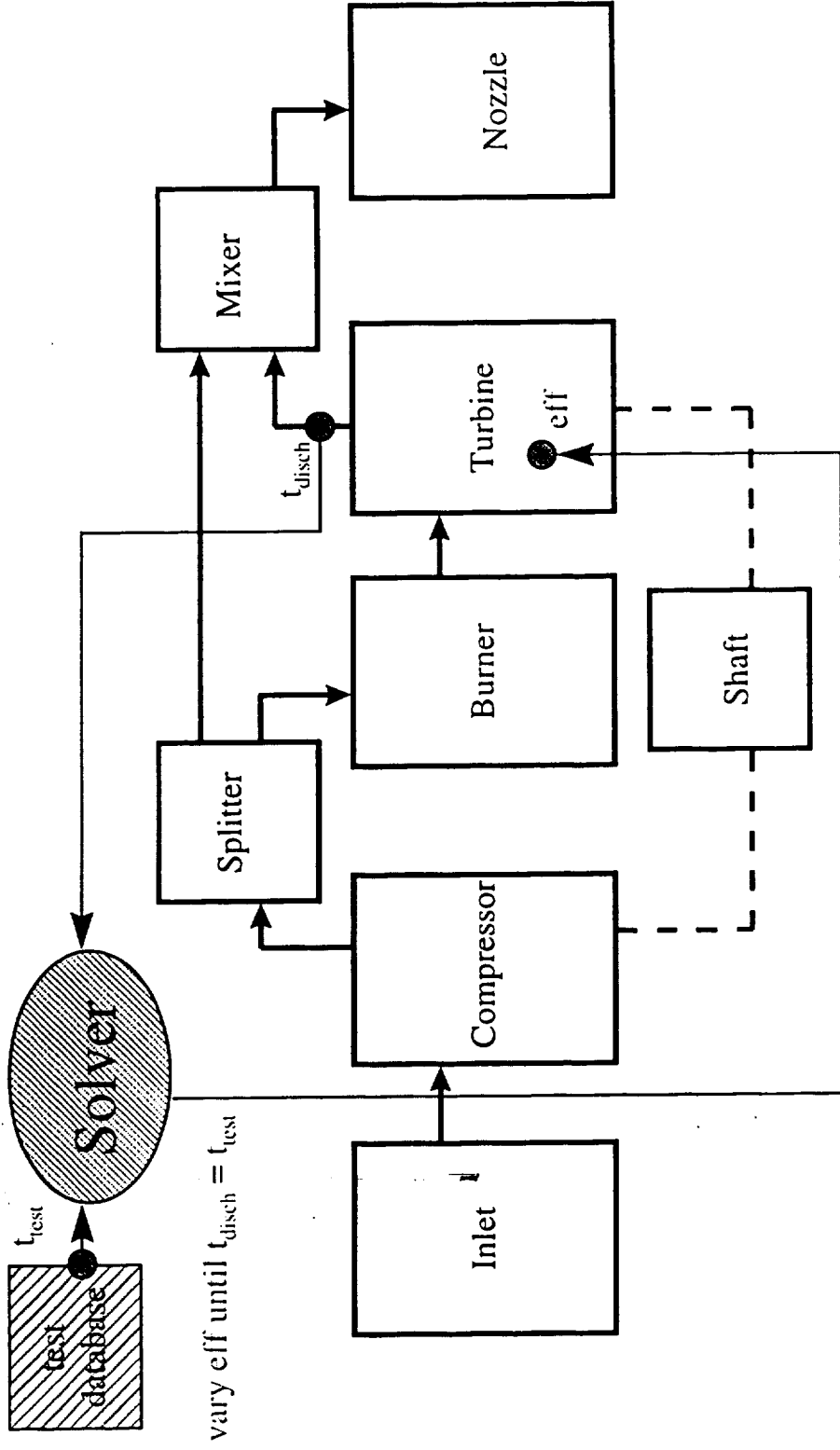
Analysis

Design

Data Reduction

Transient

Constraints



*Data Reduction*



# NCP

Analysis

Design

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thrust<sub>desired</sub>

Solver

1. vary  $w_{fuel}$  until thrust = thrust<sub>desired</sub>
2. vary  $w_{fuel}$  until  $N = N_{limit}$
3. vary  $w_{fuel}$  until temp = temp<sub>limit</sub>

Splitter

Compressor

Inlet

Burner

$w_{fuel}$

Turbine

Mixer

Nozzle

thrust

Shaft



Constraints



# NCP

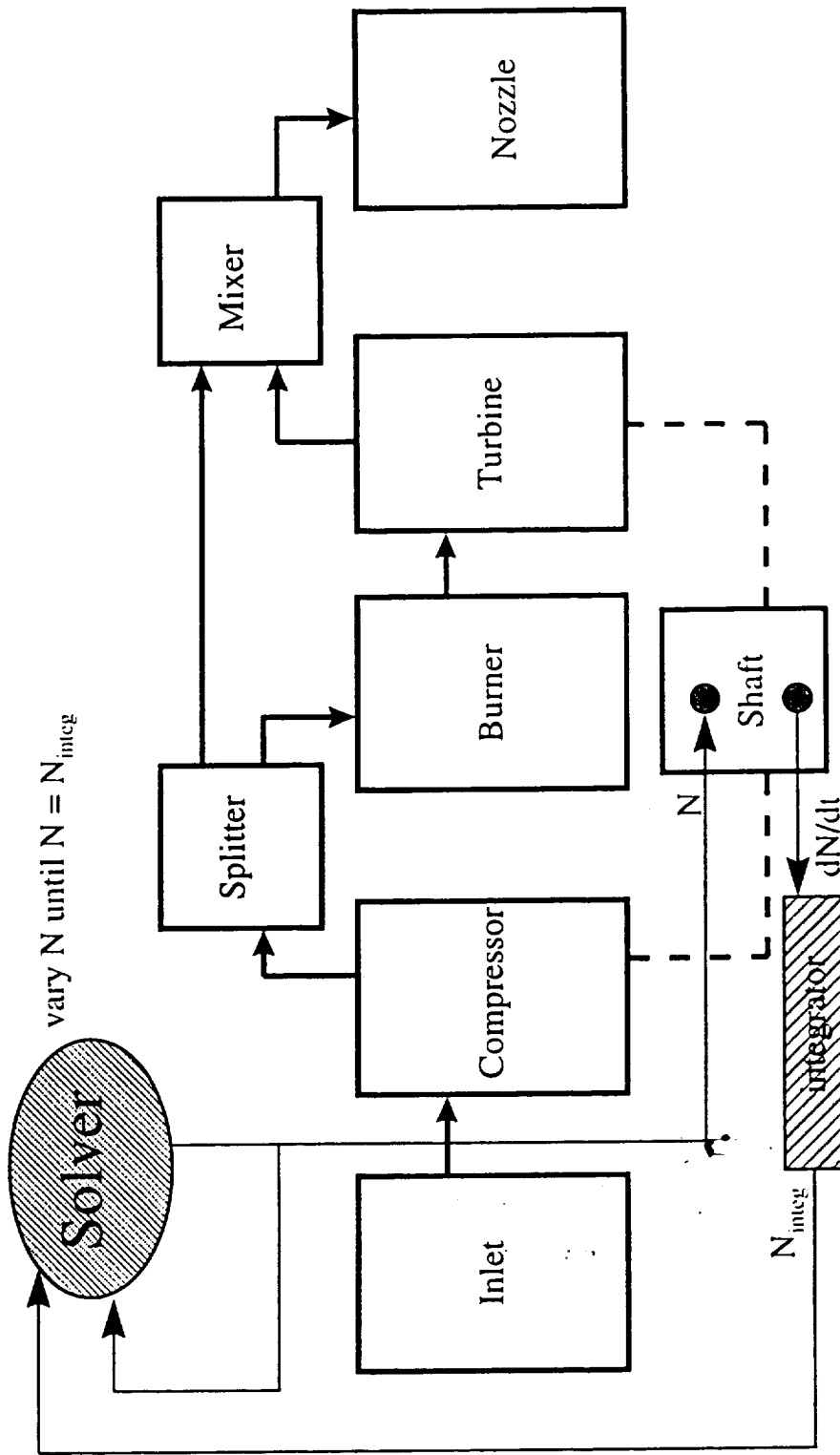
Analysis

Design

Data Reduction

Transient

Constants



*Time-Dependent*



## Special Transient Mode Functions

- Synchronization with Digital Control Elements
- Adaptive time-step ( $\Delta t$ ) computation

# NCP Time-Discrete Synchronization

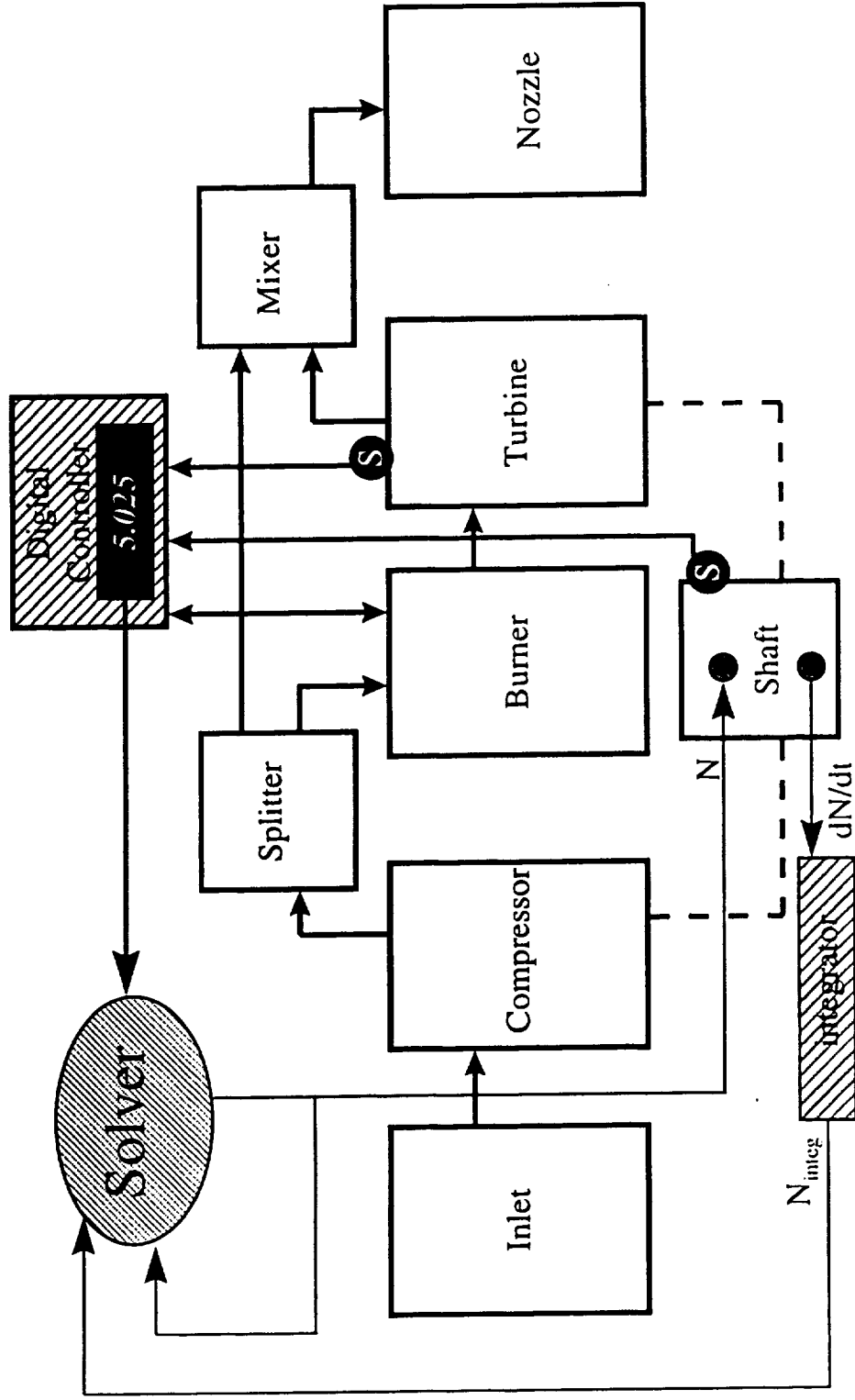
Analysis

Design

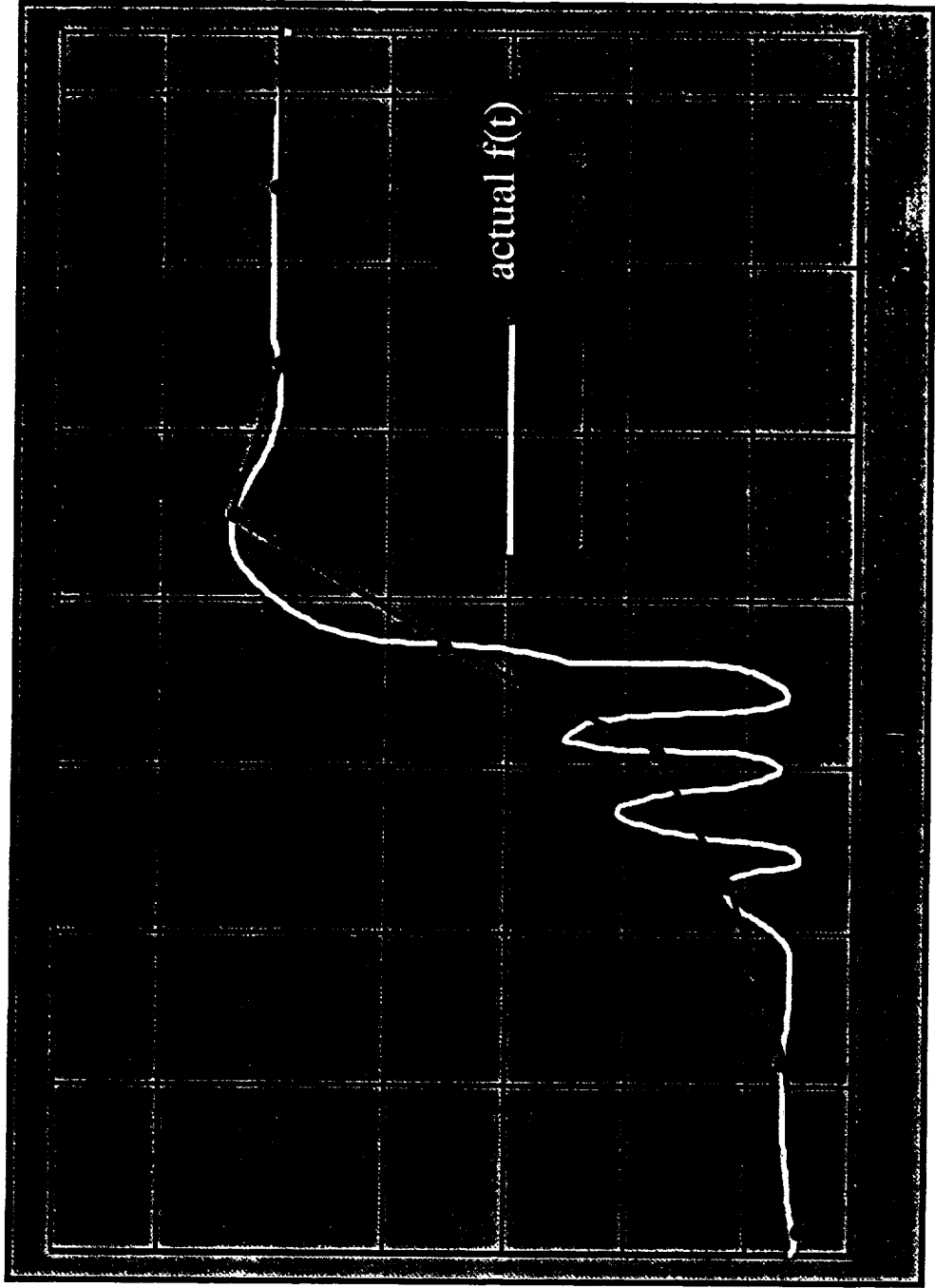
Data Reduction

Transient

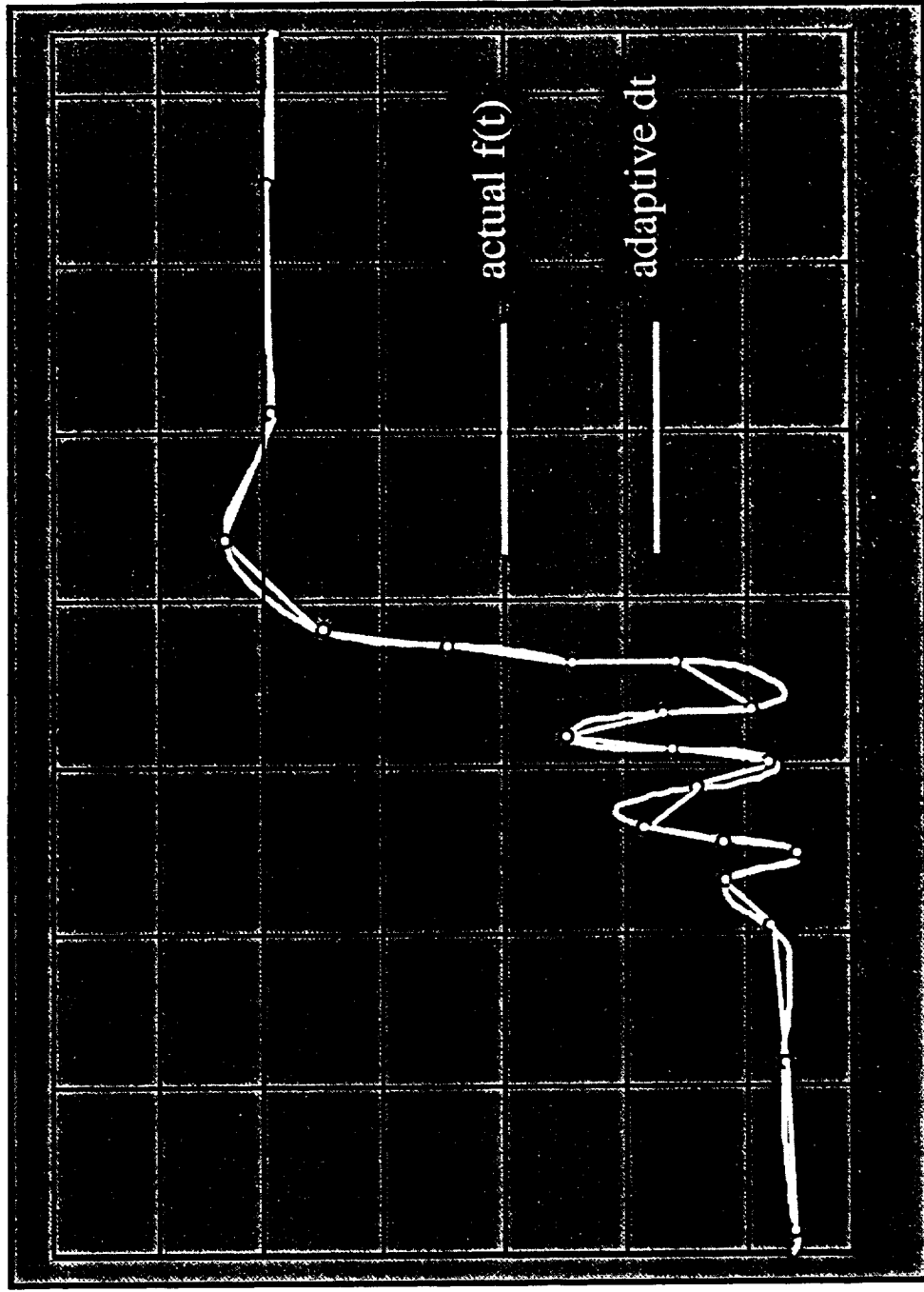
Constraints



# Adaptive Time-Stepping



# Adaptive Time-Stepping



x

## Basic Solution Methodology

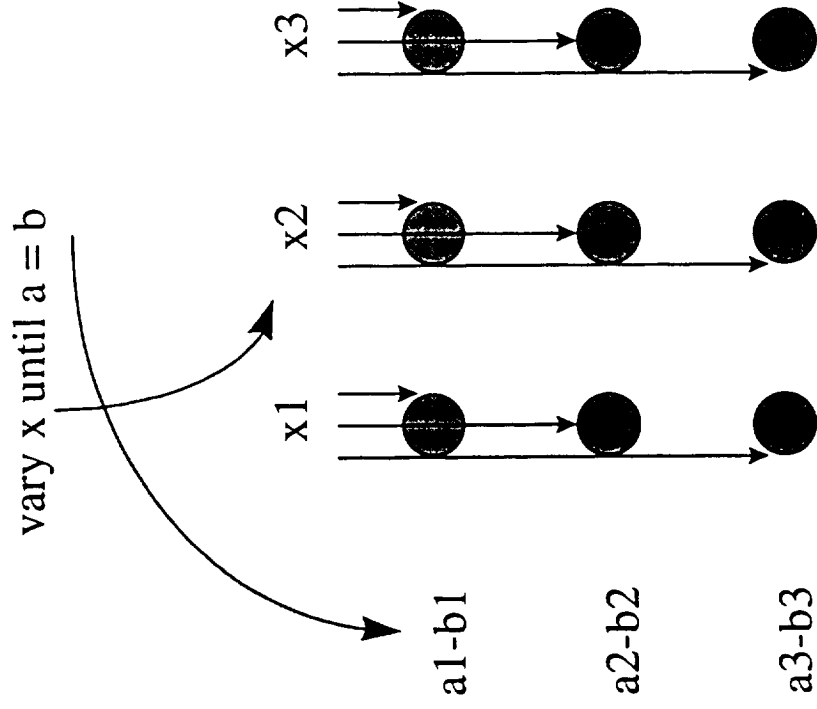
- Provides a valid physical solution at each steady-state point or transient time-step.
- Uses a Modified Newton-Raphson technique
  - *Generates matrix of partial derivatives called the Jacobian*

$$J_{ij} = \frac{\partial \bar{o}(a-b)_i}{\partial (x)_j}$$

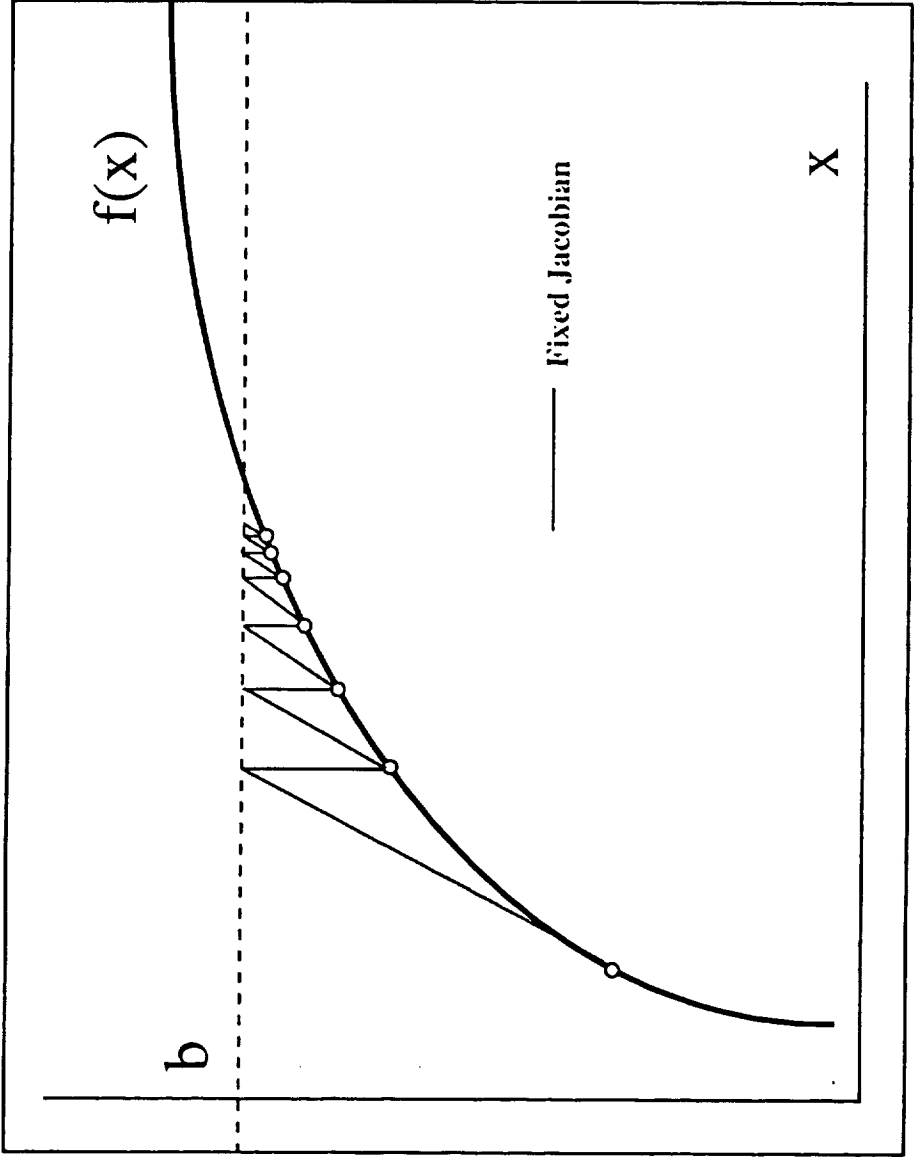
- *Inverts the matrix to compute  $x$  values that will drive errors  $(a-b)$  to zero.*
- *Modifies  $J$  and  $J^{-1}$  to account for changes in slope during convergence process.*
- *Limits the allowed change in  $x$  per iteration in order to prevent divergence.*

# Creating the Jacobian Matrix

$$J_{ij} = \frac{\partial \tilde{o}(a-b)_i}{\partial (x)_j}$$

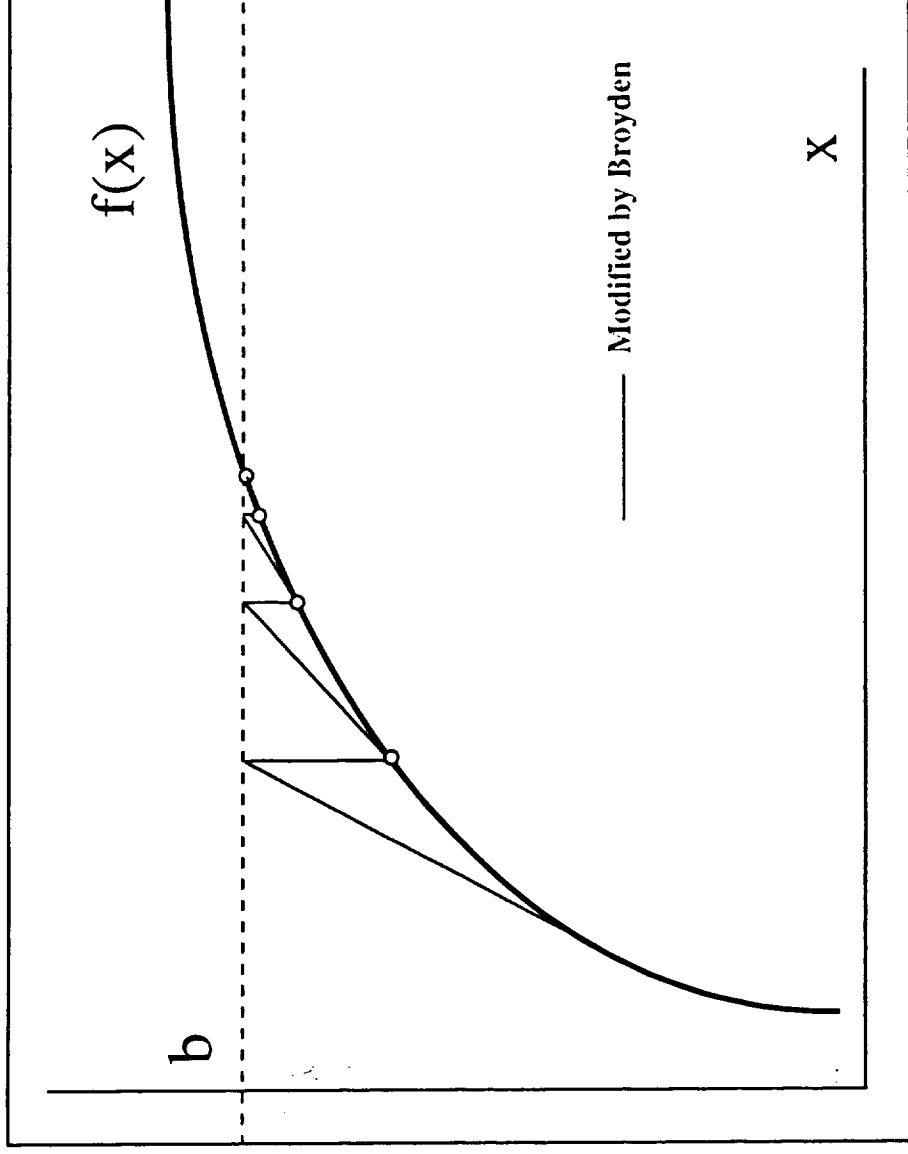


# Modified Newton-Raphson Method



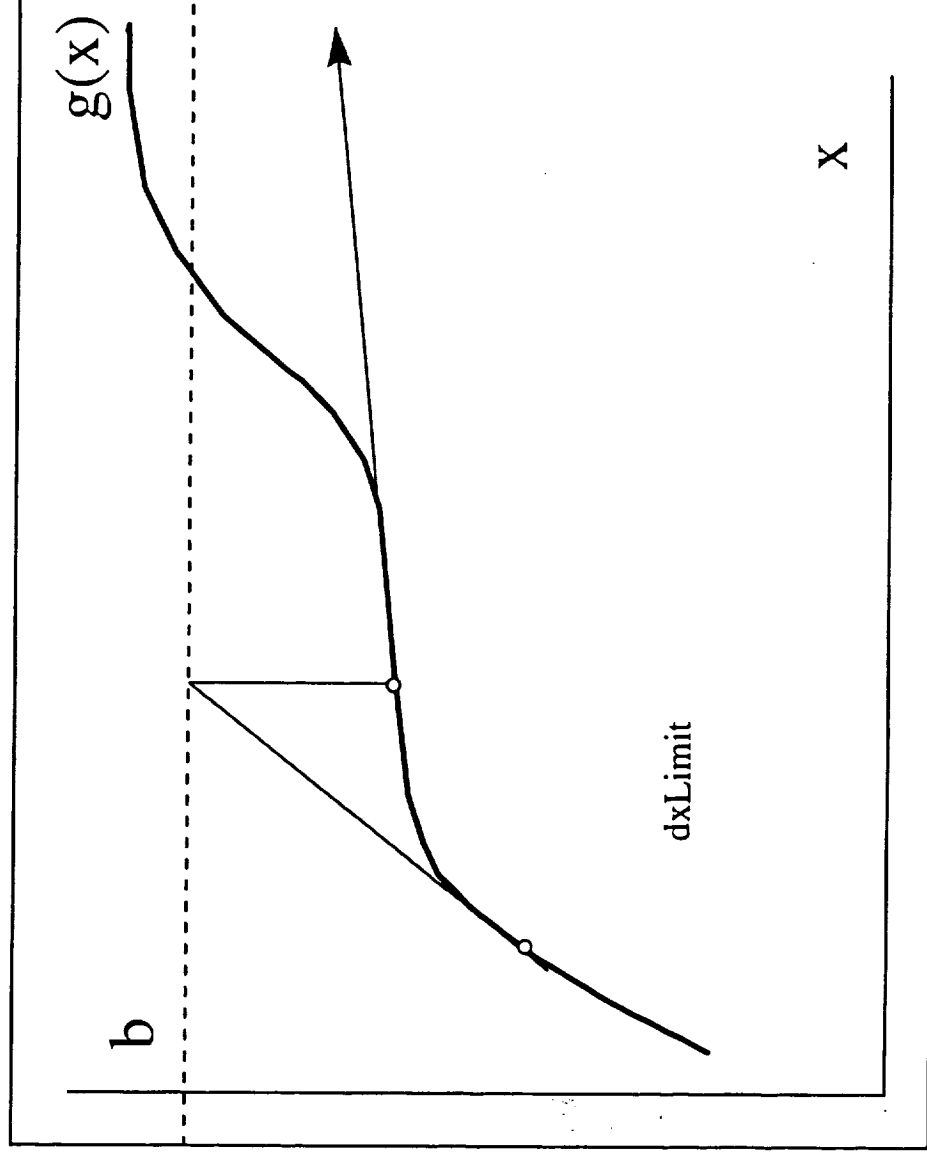
*vary  $x$  until  $f(x) = b$*

## Modified Newton-Raphson Method



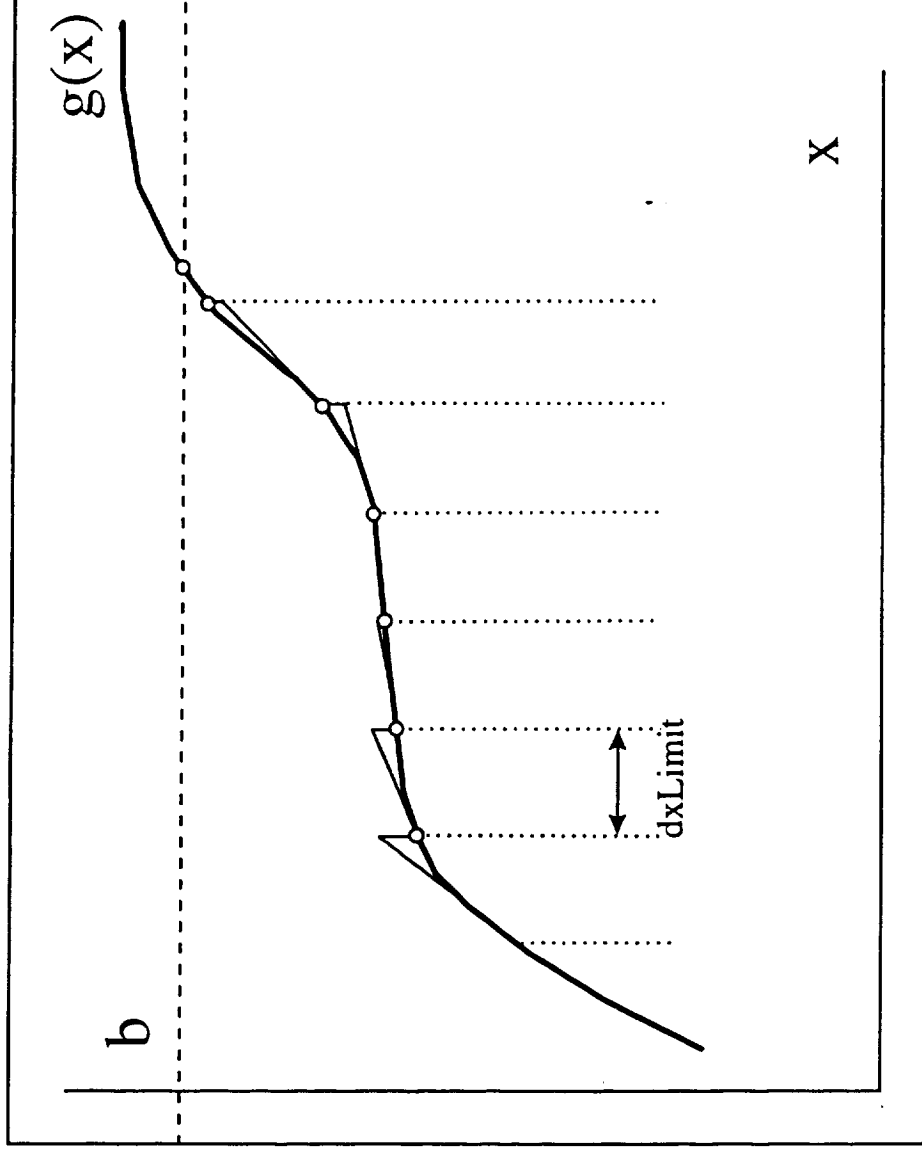
*vary  $x$  until  $f(x) = b$*

# Stiff Systems and dx-limiting



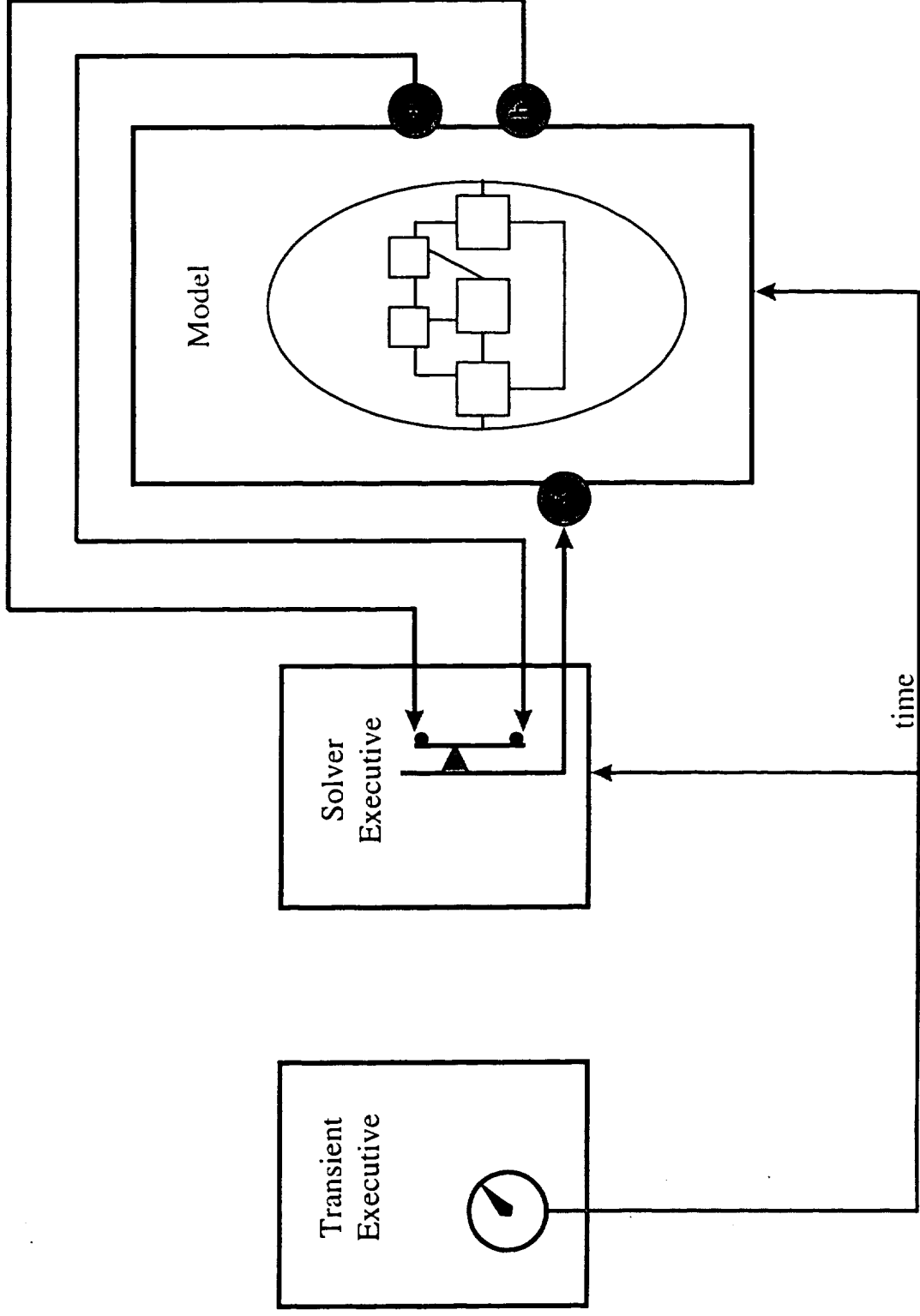
*vary  $x$  until  $g(x) = b$*

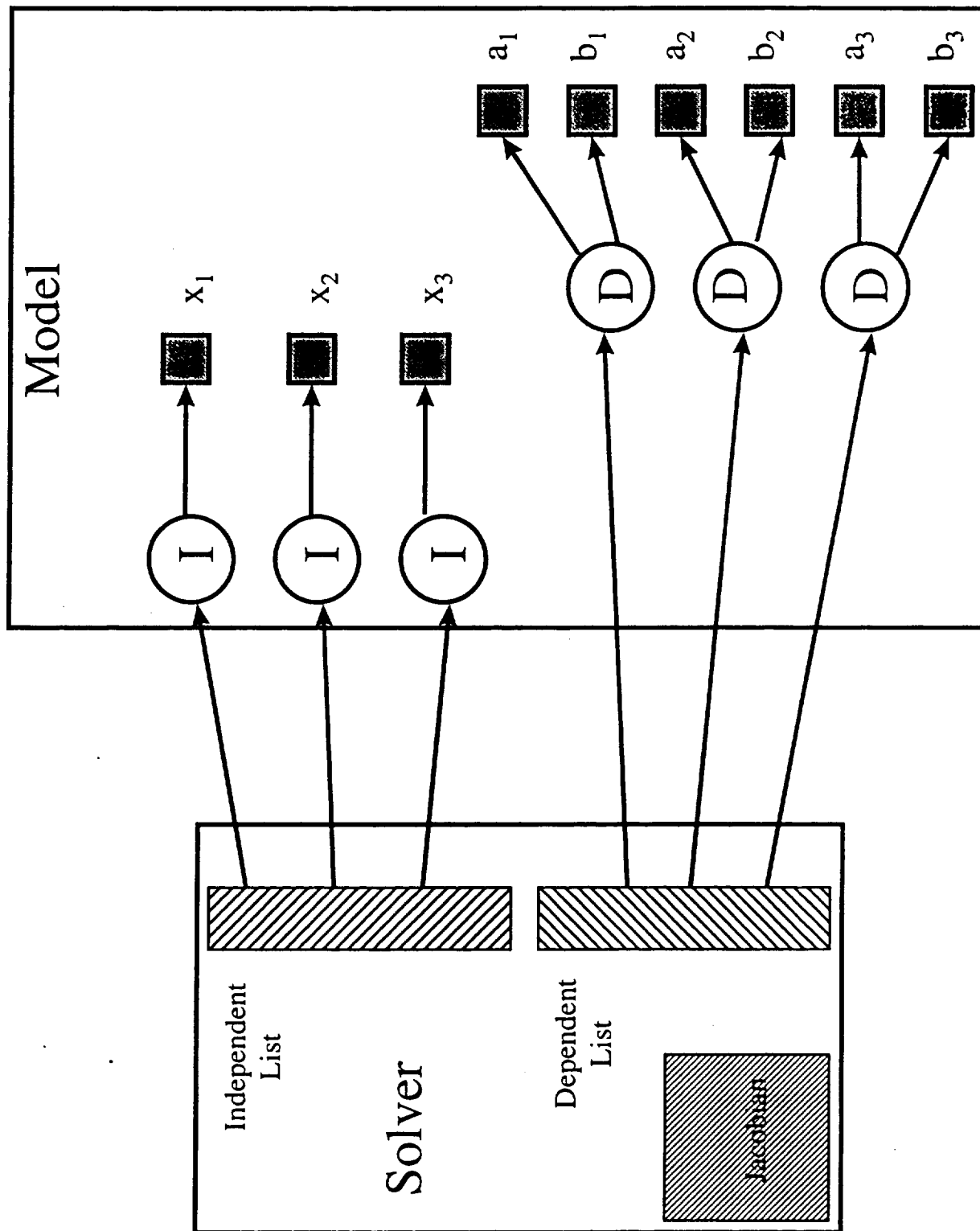
## Stiff Systems and dx-limiting



*vary  $x$  until  $g(x) = b$*

# Solver Design





## Summary

- The NCP Solver provides a great deal of versatility for system modeling of aerospace propulsion systems.
  - Analysis
  - Design
  - Data Reduction
  - Constraints
  - Transients  $\zeta$
- The Solver uses a robust solution method with an established track-record (in system modeling and finite-difference analysis).
- The Solver Subsystem has a modular, object-oriented architecture which enhances the overall flexibility and maintainability of the NCP software.